SSME EA/CIL **REDUNDANCY SCREEN**

Component Group: CIL Item:

Ducts and Lines

Part Number:

N400-01 RS007280

Component:

POGO Suppressor Accumulator

FMEA Item:

N400

Failure Mode:

Fails to contain helium/oxidizer.

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Prepared: Approved:

D. Early T. Nguyen

Approval Date: Change #: Directive #:

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| Phase | Failure / Effect Description | Criticality Hazard Reference |
|-------|--|---------------------------------|
| PSMCD | Leakage of liquid/gaseous oxygen into aft compartment and overpressurization of the aft compartment. HPOTP cavitation and overspeed. | 1 |
| 4.1 | Loss of Pogo suppression during operational phases. Loss of vehicle. | ME-C3P,D, |
| | | ME-C3S, |
| | Redundancy Screens: SINGLE POINT FAILURE: N/A | ME-C3M, |
| | | ME-C3A,C |

SSME FMEA/CIL DESIGN

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Design / Document Reference

FAILURE CAUSE: A: Parent material failure or weld failure of accumulator.

B: Parent material failure of plate.

THE ACCUMULATOR (1) IS MANUFACTURED UTILIZING INCONEL 718. INCONEL 718 WAS SELECTED FOR ITS STRENGTH, RESISTANCE TO STRESS CORROSION, CORROSION RESISTANCE, HIGH/LOW CYCLE FATIGUE CHARACTERISTICS, AND WELDABILITY (2). THE COVER PLATE (3) IS MANUFACTURED FROM 321 CRES BAR. THIS MATERIAL WAS SELECTED FOR ITS STRENGTH, FABRICABILITY, GENERAL CORROSION RESISTANCE, AND STRESS CORROSION RESISTANCE (2). MATERIALS ARE HEAT TREATED TO DEVELOP FULL MATERIAL STRENGTH AND HARDNESS. MATERIALS USED IN THE ACCUMULATOR FABRICATION ARE LOX COMPATIBLE (2). WEBS ARE USED IN THE BASE OF THE ACCUMULATOR TO DISTRIBUTE THE PRESSURE ALONG THE FLANGE AND REDUCE STRESS. MINIMUM MACHINING RADII REDUCE STRESS RISERS THAT MAY CAUSE EXCESSIVE STRAINS. SURFACE STEPS ARE USED IN AREAS WHERE INSTRUMENT PARTS AND DUCT OPENINGS ARE LOCATED TO DISTRIBUTE LOADS. A TRANSITION COLLAR BETWEEN THE INLET AND THE BALL SUPPORTS THE MOUNTING FLANGE LOADS AND PREVENTS THE NECK FROM EXPANDING UNDER INTERNAL PRESSURE. HEAT SINKS ARE USED AT THE HALF CONNECTING WELDS TO REDUCE WELD DISTORTION AND IMPROVE WELD QUALITY. MINIMUM FACTORS OF SAFETY MEET CEI REQUIREMENTS (4). HIGH AND LOW CYCLE FATIGUE LIFE FOR THE POGO ACCUMULATOR MEET CEI REQUIREMENTS (5). THE POGO ACCUMULATOR HAS COMPLETED PRESSURE CYCLING AND BURST PRESSURE DVS TESTING (6). THE ACCUMULATOR PARENT MATERIAL WAS CLEARED FOR FRACTURE MECHANICS/NDE FLAW GROWTH, SINCE THEY ARE NOT FRACTURE CRITICAL PARTS (7). TABLE 1/400 LISTS ALL THE FMEA/CIL WELDS AND IDENTIFIES THOSE WELDS IN WHICH THE CRITICAL INITIAL FLAW SIZE IS NOT DETECTABLE, AND THOSE WELDS IN WHICH THE ROOT SIDE IS NOT ACCESSIBLE FOR INSPECTION. THESE WELDS HAVE BEEN ASSESSED AS ACCEPTABLE FOR FLIGHT BY RISK ASSESSMENT (8).

(1) RS007280; (2) RSS-8582, RSS-8575; (3) RS007167; (4) RSS-8546, CP320R0003B; (5) RL00532, CP320R0003B; (6) RSS-106-23; (7) NASA TASK 117; (8) RSS-8756

SSME FML JIL **INSPECTION AND TEST**

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| Failure Causes | Significant Characteristics | Inspection(s) / Test(s) | Document Reference | |
|----------------|-----------------------------|--|---|--|
| λ, В | POGO ACCUMULATOR PLATE | | RS007280 RS007167 | |
| | MATERIAL INTEGRITY | MATERIAL INTEGRITY IS VERIFIED PER DRAWING REQUIREMENTS. | RS007280 RS007167 | |
| | | FORGING IS PENETRANT AND ULTRASONIC INSPECTED PER SPECIFICATION REQUIREMENTS. | RA0115-116 RA0115-012 | |
| | | THE FORGING TENSILE TESTS ARE VERIFIED PER SPECIFICATION REQUIREMENTS. | RB0170-153 | |
| | HEAT TREAT | HEAT TREAT IS VERIFIED PER SPECIFICATION REQUIREMENTS. | RA0611-020 | |
| | WELD INTEGRITY | ALL WELDS ARE INSPECTED TO DRAWING AND SPECIFICATION REQUIREMENTS PER WELD CLASS. INSPECTIONS INCLUDE: VISUAL, DIMENSIONAL, PENETRANT, RADIOGRAPHIC, ULTRASONIC, AND FILLER MATERIAL, AS APPLICABLE. | RL10011 RA0607-094 RA0115-116 RA0115-006 RA1115-001 RA0115-127 | |
| | ASSEMBLY INTEGRITY | ACCUMULATOR ASSEMBLY IS PROOF PRESSURE TESTED PER DRAWING REQUIREMENTS. | RS007280 | |
| | | EXTERIOR WELD IS PENETRANT INSPECTED PER SPECIFICATION REQUIREMENTS AFTER PROOF PRESSURE TEST. | RA0115-116 | |
| | FLIGHT FLOW TESTING | THE EXTERNAL SURFACE IS VISUALLY INSPECTED PRIOR TO EACH LAUNCH. | OMRSD V41BU0.03 | |
| | | A HELIUM SIGNATURE LEAK TEST IS PERFORMED PRIOR TO EACH LAUNCH. (LAST TEST) | OMRSD S00000.950 | |

Failure History:

Comprehensive failure history data is maintained in the Problem Reporting database (PRAMS/PRACA)

Reference: NASA letter SA21/88/308 and Rocketdyne letter 88RC09761.

Operational Use: Not Applicable.

SSME | _A/CIL **WELD JOINTS**

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Ducts and Lines

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N400 RS007280

Part Number:

POGO Suppressor Accumulator

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| | | | | | Root Side Not | Flaw S | al Initial Size Not ectable | |
|-------------|-------------------|-------------|-----------|-------|------------------|--------|-----------------------------------|----------|
| Component | Basic Part Number | Weld Number | Weld Type | Class | Access | HCF | LCF | Comments |
| ACCUMULATOR | RS007280 | 1 (OPT) | GTAW | ŀ | | X | Х | |
| ACCUMULATOR | RS007280 | 2-108 | GTAW | \$1 | Х | | | |